

SERVICE BULLETIN

MANDATORY

No. SB-0-002-2024-EN

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1) Planning Information

Applicability:	S/N: DF125, DF160, DF176, DF200, DF202, DF213			
Concurrent ASB/SB/SI and SL:	None			
Reason:	Yaw control path fuselage structure collision (in the case of pedals with shifted			
	position installation)			
Subject:	Perform manual repair according to instructions			
Compliance:	Immediately, before next flight			
Approval:	Inform manufacturer	Inform manufacturer		
Labour time:	0.5 hr			
Mass data:	Change of weight:	none		
	Change of CG:	none		
Electrical load data:	None			
Software modifications:	None			
References:	None			
Other Publications affected:	None			
Interchangeability of part:	None			
2) Material Information				
Material-cost and availability:	None			
Company support information:	None			
Material requirement per aircraft:	None			
Material requirement per spare part:	None			
Rework of parts:	Yes			
Special tooling and availability:	Standard sheet metal tools (metal file tools)			



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## 3) Accomplishment/Instructions

Instructions:

1) Collision takes place below and behind the instrument panel.







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2) Frame which is affected is numbered as 2\_02\_11\_09:



 In the case of shifted pedals installation (in direction to the firewall) there is a collision between the control cables and the fuselage structure.



- 4) In such a place there is a need for the cutaway to be expanded. Also check for the condition of control cables (look for any wear and fraying). Pay attention when performing the cutaway. Make sure that your tooling will not interfere with control cables. The cutaway must be performed on both sides (left and right cable).
- Enhanced hole must allow for the free movement of the cables without any further collision. Gap of 5mm between the cable and sheet metal can be considered as satisfactory.
- 6) Sharp edges must be sanded. Example of the hole enlargement:



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## Summary:

None

Inquiries:

After performing SB, it is obligatory to send the manufacturer short report accompanied by the S/N of the affected aircraft and photos of the repair.

## 1) Appendix

None

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